

# Performance Analysis of a Multiple Micro-Jet Impingements Cooling Model

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**Abstract:** The present study investigates the thermal performance of a multiple micro-jet impingements model for electronics cooling. The fluid flow and heat transport characteristics were investigated for steady incompressible laminar flow by solving three-dimensional (3D) Navier-Stokes equations. Several parallel and staggered micro-jet configurations (*ie.* inline  $2 \times 2$ ,  $3 \times 3$  and  $4 \times 4$  jets, and staggered five-jet and 13-jet arrays with the jet diameter to the channel height ratios from 0.25–0.5) were analyzed at various flow rates for the maximum temperature rise, pressure drop, heat-transfer coefficient, thermal resistance, and pumping power characteristics. The parametric investigation was carried out based on the number of jets and the jet diameters at various mass flow rates and jet Reynolds numbers. Temperature uniformity and coefficient of performance were evaluated to find out the trade-off among the various designs investigated in the present study. The maximum temperature rise and the pressure drop decreased with an increase in the number of jets except in the case of staggered five-jet array. A higher temperature uniformity was observed at higher flow rates with a decrease in the coefficient of performance. The performance parameters, such as thermal resistance and pumping power, showed a conflicting nature with respect to design variables (*viz.* jet diameter to stand-off ratio and interjet spacing or number of jets) at various Reynolds numbers within the laminar regime.

**Keywords:** Jet impingement, Thermal resistance, Pumping power, Temperature uniformity, Heat transfer coefficient, Coefficient of performance.

## تحليل الأداء لنموذج التبريد بالارتطام متعدد النافثات الدقيقة

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**المخلص:** تبحث هذه الدراسة الأداء الحراري لنموذج الارتطام متعدد النافثات الدقيقة والمستخدم في تبريد الالكترونيات. تم تحري خواص المائع المستخدم والانتقال الحراري في حالة الجريان الصفحي الثابت والغير قابل للانضغاط وذلك عن طريق حل معادلة نافيه ستوكس ثلاثية الأبعاد. كما بحثت الدراسة تأثير ترتيب النافثات المتوازية والمتداخلة على الأداء باستخدام ترتيب خطي على شكل  $2 \times 2$  و  $3 \times 3$  و  $4 \times 4$ ، وترتيب تداخلي باستخدام 5 نافثات و 13 نافثة، وباستخدام نسبة قطر النافثة لعمق القناة ما بين 0.25 إلى 0.5، وتحليلها باستخدام معدلات تدفق مختلفة لأقصى ارتفاع لدرجة الحرارة، وانخفاض الضغط، ومعامل الانتقال الحراري، والمقاومة الحرارية، وقدرة الضخ. كما تم تقييم مدى توحيد درجة الحرارة ومعامل الأداء لمعرفة المفاضلة بين مختلف التصميم التي حققت فيها رينولد المتباينة. إن أقصى ارتفاع لدرجة الحرارة وانخفاض الضغط وجد أنها تقل مع زيادة عدد النافثات باستثناء الترتيب التداخلي ذي الخمس نافثات. ولوحظ وجود أعلى انتظام في درجة الحرارة مع معدلات التدفق العالية المرافق لها انخفاض في مستوى الأداء. أن معايير الأداء، مثل المقاومة الحرارية وقدرة الضخ أظهرت تعارض مع متغيرات التصميم كنسبة قطر النافثات لمعامل المكث، والتباعد بين النافثات، وعدد النافثات، وذلك على مختلف أرقام رينولد في الجريان الصفحي.

**الكلمات المفتاحية:** ارتطام النافثة، المقاومة الحرارية، قدرة الضخ، انتظام درجة الحرارة، معامل الانتقال الحراري، معامل الأداء.

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## Nomenclature

$A_s$	Surface area of the substrate base, m <sup>2</sup>
$C_p$	Specific heat at constant pressure, J kg <sup>-1</sup> K <sup>-1</sup>
$COP$	Coefficient of performance
$d$	Diameter of the nozzle, m
$H$	Height of the channel, m
$h$	Heat transfer coefficient, W m <sup>-2</sup> K <sup>-1</sup>
$k$	Thermal conductivity, W m <sup>-1</sup> K <sup>-1</sup>
$l$	Length of the nozzle, m
$l_x, l_y, l_z$	Length, width and height of the silicon substrate, respectively, m
$N$	Number of observations for calculating the mean value
$n$	Number of jets
$p, \Delta p$	Pressure and pressure-drop, respectively, Pa
$P$	Pumping power, W
$P''$	Pumping power flux, W m <sup>-2</sup>
$q$	Heat flux, W m <sup>-2</sup>
$R_{th}$	Thermal resistance, K W <sup>-1</sup>
$S$	Interjet spacing, m
$T, \Delta T$	Temperature and temperature-rise, respectively, K
$t_s$	Thickness of the substrate base.
$u, v, w$	Orthogonal components of fluid velocity, m s <sup>-1</sup>
$V$	Velocity vector
$\dot{V}$	Volume flow rate, m <sup>3</sup> s <sup>-1</sup>
$x, y, z$	Orthogonal coordinate system
$\dot{m}$	Mass flow rate, kg s <sup>-1</sup>

### Greek symbols

$\mu$	Dynamic viscosity, kg s <sup>-1</sup> m <sup>-1</sup>
$\rho$	Density, kg m <sup>-3</sup>
$\sigma$	Standard deviation, K

### Subscripts

4,5,9,13,16	Number of jets
$avg$	Area-average value
$f$	Fluid
$i$	Inlet
$j$	Jet
$max$	Maximum value
$mean$	Mean value
$\bar{n}$	Vector in wall/plane normal direction
$o$	Outlet
$s$	Substrate

## 1. Introduction

The growing demand for higher heat flux removal and the occurrence of hot spots from ultra-large scale integrated (ULSI) circuits of power electronics, concentrated photovoltaic, and high-power light emitting diodes (LEDs) have posed a major challenge to heat transfer researchers. The sensitivity, reliability, performance and efficiency of electronic devices are highly temperature dependent and a high temperature may lead to irreparable loss of electronics (Hamann *et al.* 2007). The increase in junction temperature decreases the life span of LEDs exponentially (Narendran and Gu 2005). The heat generated at the P-N junction is transferred to the substrate material which must be dissipated to allow proper functioning of electronics. These challenges have encouraged researchers to develop new, alternative cooling techniques at the mini- and micro-levels. As the air cooling techniques reach their limits, the liquid flow through parallel micro-channels provides a potential solution for the high heat-flux management of electronics. Tuckerman and Pease (1981) were the first to examine liquid-flow micro-channel heat sinks. The liquid flows from one end to the other through narrow parallel channels, leaving higher temperatures at the inlet side and lower temperatures at the outlet side (Husain and Kim 2008a; Kawano *et al.* 1998; Knight *et al.* 1992; Liu and Garimella 2005; Toh *et al.* 2002; Qu and Mudawar 2002).

The hot-spot occurrence and high temperature-gradients lead to temperature non-uniformity on the heated surface and, consequently on the chip surface, inducing thermal stresses which eventually reduce the life span of the electronics. Jet impingement cooling offers an efficient solution for high heat-flux management of an entire surface as well as hot spots in power electronics with relatively more uniform temperature distribution over the chip surface. The recent developments in electronics cooling have directed researchers' endeavors towards achieving higher temperature uniformities as well as hot-spot management by the application of micro-jet impingements on the heated substrate (Lee and Vafai 1999; Wu *et al.* 1999). Wu *et al.* (1999) examined experimentally the heat transfer characteristics of 500  $\mu\text{m}$  and 550  $\mu\text{m}$  single jets as well as a jet-array with  $H$  ranging from 200–3000  $\mu\text{m}$ . They obtained an area-averaged heat transfer coefficient of 320  $\text{W}/\text{m}^2\text{K}$  for a 500  $\mu\text{m}$  single jet with  $H = 750 \mu\text{m}$  at a pressure-drop of about 34.5 kPa. They also reported a higher cooling efficiency at a lower driving pressure, which

decreased with further increase in the driving pressure. A comparative analysis of the micro-channels and jet impingement cooling shows that the micro-channel cooling is preferable for a heated surface size smaller than 70 mm  $\times$  70 mm; and the jet impingement is comparable to or better than the micro-channel cooling for a large target plate with a proper arrangement of spent flow after impingements (Lee and Vafai 1999).

The potential of micro-jet impingement cooling in electronics has motivated researchers to examine micro-jet impingements cooling models both numerically and experimentally. Moreover, with the advent of advanced fabrication processes, the production of micro-scale jets has become more convenient than before (Jackson 2006). Wang *et al.* (2004) investigated experimentally single- and two-phase flow micro-jet impingement cooling systems. They fabricated a single and multi-jet array with circular orifice diameters ranging from 40–76  $\mu\text{m}$ , which removes 90 W using a four-jet array at a flow rate of eight ml/min with a temperature rise of 100  $^{\circ}\text{C}$  for a 1 cm  $\times$  1 cm chip. Fabbri and Dhir (2005) examined impinging jet arrays in a circular pattern with three different radial and circumferential pitches of 1, 2 and 3 mm. They investigated free jets of both water and FC-40 at Reynolds numbers ranging from 73–3813, and reported heat transfer coefficients ranging from 6–60  $\text{kW}/\text{m}^2\text{K}$ . Sung and Mudawar (2006) proposed a hybrid cooling scheme for electronics cooling, which combines the cooling benefits of both microchannel and jet impingement. They carried out experimental and numerical analyses of slot-jet impingements in the microchannels which showed low rise and small temperature gradients across the surface.

Michna *et al.* (2009) experimentally investigated an array of micro-jets for stagnation heat transfer and obtained a maximum  $11 \times 10^6 \text{ W}/\text{m}^2$  heat flux for the rise of a surface temperature less than 30 $^{\circ}\text{C}$ . They observed that the Reynolds number, the Prandtl number, and the area ratio (total area of jets divided by the surface area) can have significant effects on heat transfer. Browne *et al.* (2010) reported micro-jet arrays of jet diameters 52  $\mu\text{m}$  and 112  $\mu\text{m}$  and a jet spacing of 250  $\mu\text{m}$  for heat transfer characteristics. They observed similar trends for micro-jet arrays for conventional- and macro-sized jet arrays. Husain *et al.* (2013a) investigated multiple micro-jet impingement heat sinks and carried out parametric optimization to find out optimal designs. In other studies, Husain *et al.* (2013b; 2013c) proposed turbulent and laminar flow numerical optimization models of jet impingement system for high power LEDs and

ULSI circuits cooling, respectively. Wang and Peles (2014) investigated heat transfer enhancement through passive and active flow control by a jet issued from a micro pillar in a channel. The combined effect of micro-jet and pillar-enhanced spatially-averaged the Nusselt number by 80%.

Macro-jet impingement cooling has been the subject of intense research with its applications in macro-heat sinks and turbomachines before the micro-jet impingement was applied in electronics cooling and micro-systems. The macro-jet impingement has a rich history, with most of the investigations based on air jet impingements (Garimella and Rice, 1995; Webb and Ma 1995; Womac *et al.* 1993; Womac *et al.* 1994;). Womac *et al.* (1993) proposed correlations for heat transfer and concluded that the  $H/d$  ratio has a negligible effect on heat transfer, but an increase in velocity improves heat transfer. Garimella and Rice (1995) characterized a single submerged and confined jet for local heat transfer. They examined the effects of  $d$ ,  $H/d$ ,  $l/d$ , and the flow rate on heat transfer and proposed correlations based on the jet Reynolds number. Heo *et al.* (2011) investigated an inclined elliptic jet for cross-flow analysis through numerical methods, observing that the size of the recirculation region just upstream of the jet and the length of the potential core of the jet significantly affect the heat transfer process.

In view of the limited literature on the liquid flow multiple micro-jet impingement cooling and optimal jet-array configuration,, the present study is devoted to investigating thermal performance and carries out parametric analyses of a liquid-flow multiple micro-jet impingements cooling model built on the backside of a silicon chip substrate. The effectiveness of several multi-jet configurations (*ie.* inline  $2 \times 2$ ,  $3 \times 3$ , and  $4 \times 4$  jets and staggered five-jet and 13-jet arrays with nozzle diameters of,  $50 \mu\text{m}$ ,  $76 \mu\text{m}$ , and  $100 \mu\text{m}$ ) were analyzed at different flow rates to determine the thermal and pressure-drop characteristics.

## 2. Mathematical Model and Simulation Strategy

### 2.1 Geometrical Model

A multiple micro-jet impingements cooling model consisting of a fluid channel can be fabricated on a silicon substrate and micro-nozzles on a cover plate (Fig. 1). The design can be implemented at the backside of a silicon chip that emits heat flux from the integrated circuitry. The nozzles were designed on the substrate, and a water channel was created to allow liquid to pass

through the nozzles as discussed by Wang *et al.* (2004). The liquid jets take the heat while striking at the heated surface. The flow then turns to the radial direction and reaches the outlet ports (Fig. 1). For a conservative analysis, the substrate on which these nozzles were designed was not taken for analysis, whereas the nozzles with a fluid channel and substrate base were included in the computational domain (Fig. 1(b)). The nozzles and outlet channel of the substrate can be fabricated separately and bonded together with epoxy to form the designed structure as discussed by Wang *et al.* (2004).

The overall dimensions of the micro-jet impingements domain (Fig. 1) are  $10 \text{ mm} \times 10 \text{ mm} \times 0.6 \text{ mm}$ . The thickness of the substrate base is  $100 \mu\text{m}$  and the height of the liquid channel is  $200 \mu\text{m}$ . The thickness of the nozzle plate in which multiple nozzles are designed is  $300 \mu\text{m}$ . One half of the entire domain was taken as the computational domain due to the symmetry about the central plane (Fig. 1). The various parameters that affect the performance of the model are the thickness of the substrate base ( $t_s$ ), nozzle diameter ( $d$ ), length of the nozzle ( $l$ ), interjet spacing ( $S$ ), number of nozzles ( $n$ ), channel height ( $H$ ), and the distribution of the jets (*ie.* the inline and staggered arrays) (Fig. 2). The geometric parameters,  $l$  and  $t_s$ , were kept constant due to their unidirectional effects on the model performance (Liu and Garimella 2005; Michna *et al.* 2009), whereas the channel height ( $H$ ) was kept constant to keep concise the parametric investigations.

### 2.2 Mathematical Model and Boundary Conditions

Three-dimensional (3D) numerical analyses were performed for steady state incompressible laminar flow and conjugate heat transfer. The coolant, deionized ultra-filtered (DIUF) water, was allowed to flow through the micro-nozzles and channel. The governing equations of the model (*viz.* conservation of mass, momentum and energy) for fluid flow and conjugate heat transfer can be written in vector form as follows:

$$\nabla \cdot (\rho_f \mathbf{V}) = 0 \quad (1)$$

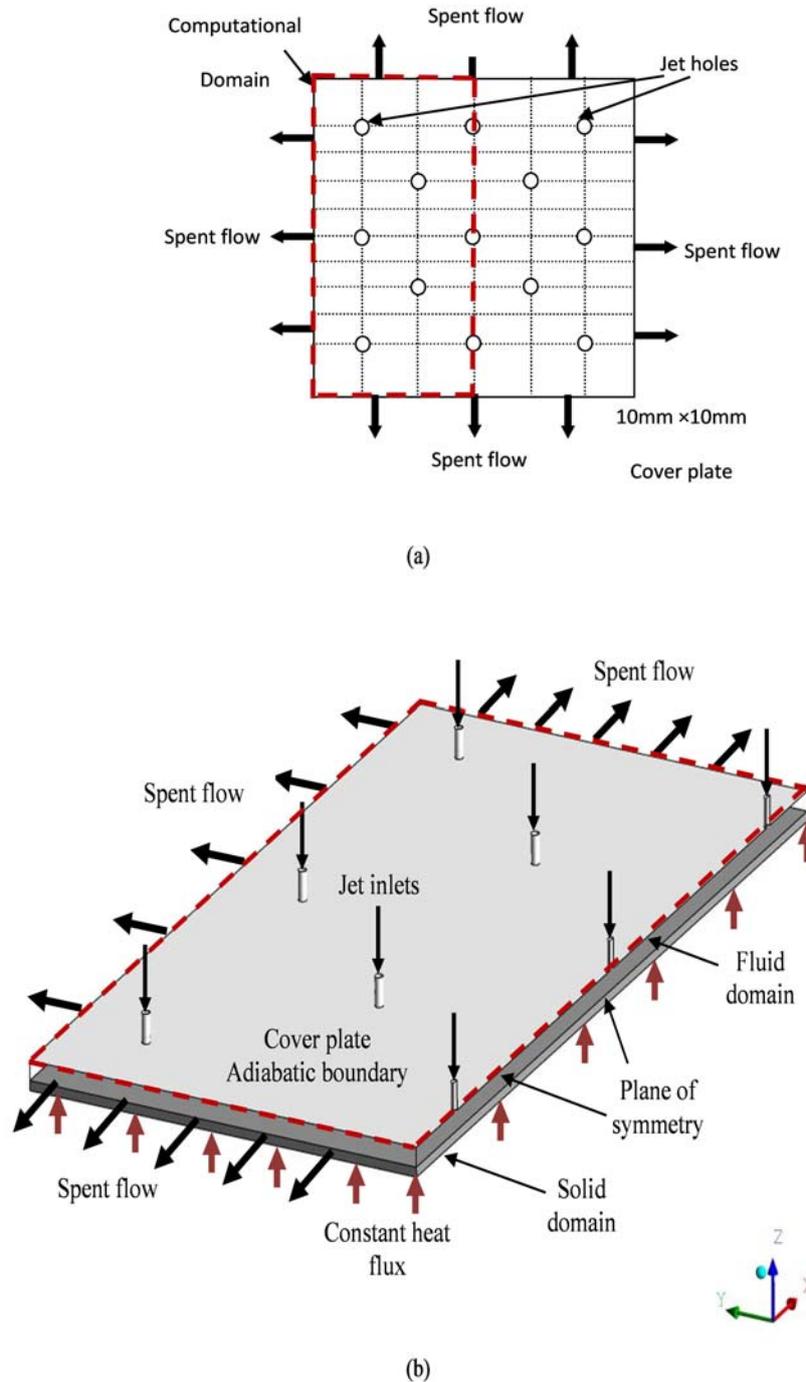
$$\mathbf{V} \cdot \nabla (\rho_f \mathbf{V}) = -\nabla p + \nabla \cdot (\mu_f \nabla \mathbf{V}) \quad (2)$$

For the fluid

$$\mathbf{V} \cdot \nabla (\rho_f C_{p,f} T_f) = \nabla \cdot (k_f \nabla T_f) \quad (3)$$

For the substrate

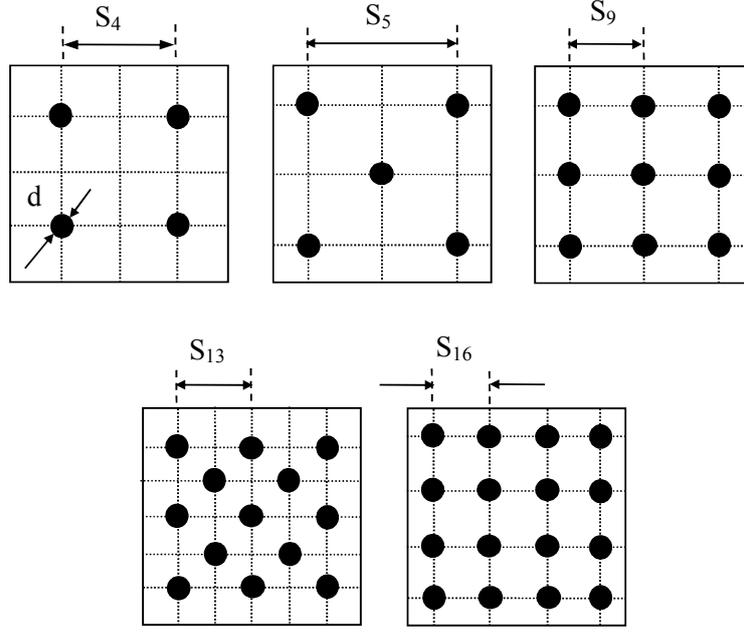
$$\nabla \cdot (k_s \nabla T_s) = 0 \quad (4)$$



**Figure 1.** Schematic of multiple micro-jet-impingements model. (a) Cover plate with jet holes, and (b) Computational domain with boundary conditions.

The numerical simulations were carried out using a commercial computational fluid dynamics (CFD) code that employs coupled algebraic multi-grid method (CFX Theory 2006; Raw 1996). The

thermophysical properties (Incropera and DeWitt 2002) of the coolant DIUF water were allowed to vary with temperature and updated after every iteration to take micro-scale effects into account



**Figure 2.** Various micro-jet configurations on the nozzle plate. (a) Inline  $2 \times 2$  jet array,  $S_4 = 5$  mm, (b) Staggered five-jet array,  $S_5 = 7.0$  mm, (c) Inline  $3 \times 3$  jet array,  $S_9 = 3.5$  mm, (d) Staggered 13-jet array,  $S_{13} = 4$  mm, and (e) Inline  $4 \times 4$  jet array,  $S_{16} = 2.5$  mm

(Herwig and Mahalikar 2006; Husain and Kim 2008b; Toh *et al.* 2002) during the iterative solution. A non-uniform hexahedral mesh was used in a specified computational domain to implement a numerical scheme (Fig. 3(a)). A high resolution scheme was applied to discretize advection terms of the governing equations. The top and bottom faces of the fluid domain interfaced with the substrate base and cover plate, respectively. The interior walls of the nozzles were assigned with adiabatic boundary conditions along with a no slip ( $V = 0$ ) boundary condition. For conservative analysis, a constant heat flux was assigned as a heat source at the base of the substrate. A no-slip condition ( $V = 0$ ) was applied at the interior walls of the nozzles and channel.

The thermal boundary conditions are:

$$-k_s \frac{\partial T}{\partial z} = q \text{ at } z = 0, \quad -\frac{l_x}{2} \leq x \leq \frac{l_x}{2}, \text{ and } 0 \leq y \leq \frac{l_y}{2}$$

$$-k_s \frac{\partial T}{\partial z} = 0 \text{ at } -\frac{l_x}{2} \leq x \leq \frac{l_x}{2}, \text{ and } 0 \leq y \leq \frac{l_y}{2}$$

$$-k_s \frac{\partial T}{\partial n} = 0 \text{ at the nozzle walls.}$$

The boundary conditions on the symmetric plane are as follows:

$$-k_s \frac{\partial T}{\partial n} = 0, \quad v = 0, \quad \frac{\partial u}{\partial n} = 0, \quad \frac{\partial p}{\partial n} = 0, \text{ and } \frac{\partial w}{\partial n} = 0$$

For the solid-liquid interface, the conditions are

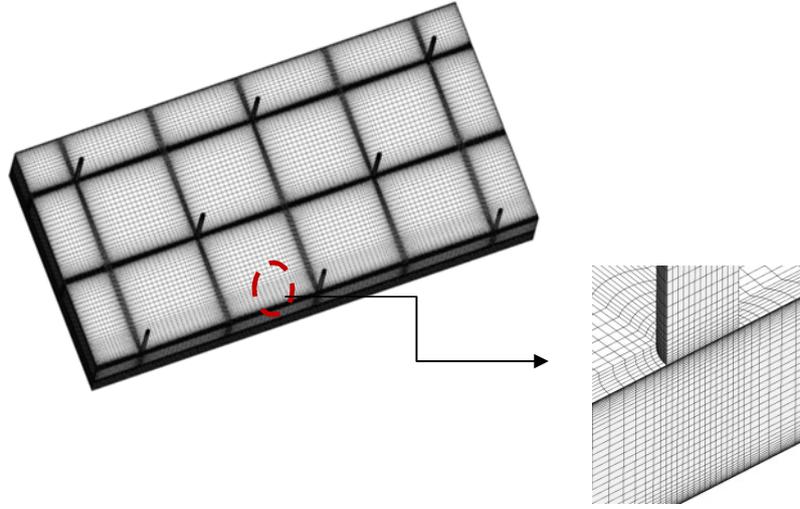
$$T_s = T_f \text{ and } -k_s \frac{\partial T_s}{\partial n} = -k_f \frac{\partial T_f}{\partial n}$$

The overall thermal resistance is defined as

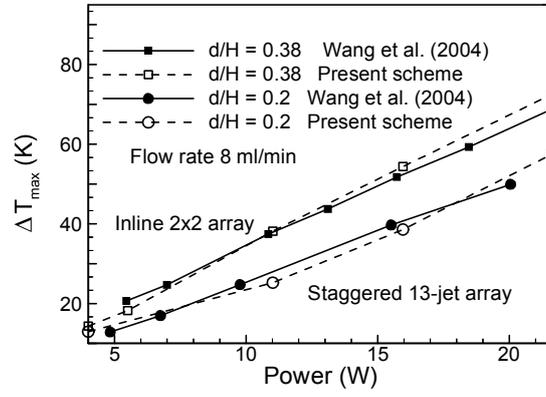
$$R_{th} = \frac{\Delta T_{\max}}{qA_s} \quad (5)$$

The maximum temperature-rise in the substrate is defined as

$$\Delta T_{\max} = T_{s,\max} - T_{f,i} \quad (6)$$



(a)



(b)

**Figure 3.** (a) Unstructured hexahedral grid, and (b) Comparison of numerical results with the results of Wang *et al.* (2004) for an inline  $2 \times 2$  array and staggered 13-jet array.

The pumping power required to drive the fluid through the model can be evaluated as follows:

$$P = \dot{V} \cdot \Delta p \quad (7)$$

The pressure-drop is defined as:

$$\Delta p = p_o - p_i \quad (8)$$

The Reynolds number at the jet inlet was defined as:

$$Re = \frac{\rho V_{j,i} d_n}{\mu} \quad (9)$$

where  $V_{j,i}$  is the velocity at the inlet. The heat transfer coefficient was evaluated as

$$h = \frac{q}{(T_{j,i} - T_s)} \quad (10)$$

The coefficient of performance (COP) can be represented as the ratio of the heat flux to the pumping-power flux:

$$COP = \frac{q}{P''} \quad (11)$$

where  $P''$  is defined as  $P'' = \frac{P}{A_j}$ .

The degree of temperature uniformity at the impingement surface can be assessed in terms of the standard deviation ( $\sigma$ ) as follows:

Temperature uniformity,

$$\sigma = \sqrt{\sum_{i=1}^N (T_i - T_{mean})^2} \quad (12)$$

where  $T_1, T_2, T_3, \dots, T_N$  are the temperatures at  $N$  sample points on the surface, and  $T_{mean}$  is the mean surface temperature. The temperature uniformity ( $\sigma$ ) at the surface increased with a decreasing  $\sigma$  value.

For a conservative analysis and simplification of the numerical model, radiation effects were ignored. The convergence was checked by the residual target of  $10^{-6}$ .

### 3. Results and Discussion

The grid dependency tests were carried out for both solid and liquid domains, and a grid of approximately 1,000,000 elements was used for a typical design of  $d/H = 0.33$  in an inline  $2 \times 2$  jet-array. For a grid of 500,000 elements, the changes in pressure-drop ( $\Delta p$ ) and maximum temperature-rise ( $\Delta T_{max}$ ) were more than 4%, whereas the changes in  $\Delta p$  and  $\Delta T_{max}$  were less than 2% for grid systems of 1,500,000 and 2,000,000 elements. The validation of the present numerical scheme for a laminar flow in a channel and jet impingement, respectively, was reported in the authors' previous works (Husain and Kim 2008a; Husain *et al.* 2013a). Further validation of the micro-jet impingement system is presented in Fig. 3(b) for an inline  $2 \times 2$  jet-array and staggered 13-jet array. The numerical reproduction of the data showed reasonable agreement with the experimental results reported by Wang *et al.* (2004).

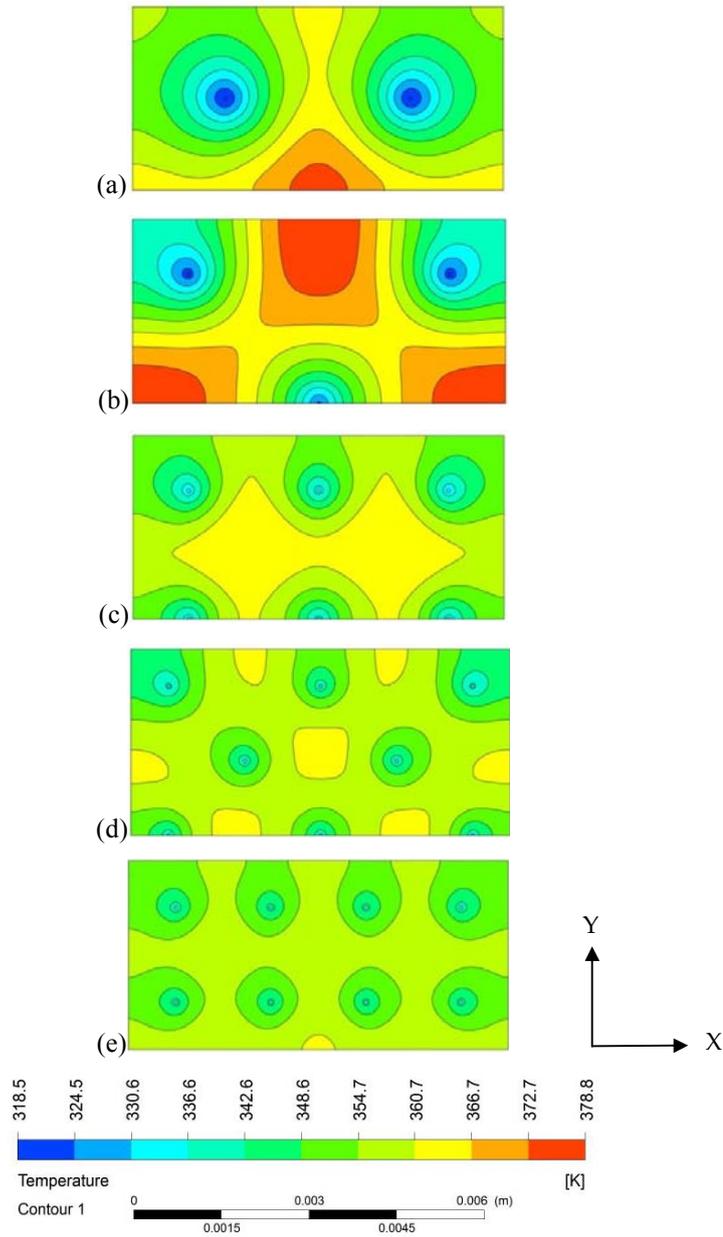
The temperature distribution and uniformity on the heated surface were presented by local temperature plots and the overall standard deviation of the temperature at the solid-fluid interface (Fig 3). Figure 4 shows the temperature distribution at the impingement surface at a constant mass flow rate. A substantially small temperature range and higher temperature uniformity were observed for the design with the higher number of jets. The high temperature zones were present in the interjet spacings away from the stagnation region. With the increased number of jets, the interjet spacing decreased, resulting in high

fluid mixing and the occurrence of secondary flow. The strength of the secondary flow increased with the further decrease in the interjet spacing. The increase in number of impingements resulted in increased temperature uniformity and decreased temperature-gradient monotonously. The maximum-temperature-rise increases with the increase in jet diameter to the channel height ratio (Fig. 5) shows the temperature distribution and uniformity at the impingement surface for  $d/H = 0.2-0.5$  and  $\dot{m}=5.32 \times 10^{-4} \text{ kg.s}^{-1}$ . At a constant mass flow rate, the jet velocity and jet Reynolds number decreased with the increase in  $d/H$  ratio. The reduced jet Reynolds number led to a decrease in the heat transfer coefficient and an increase in the  $\Delta T_{max}$  (Fig. 5).

The performance of the various jet configurations (*ie.* inline  $2 \times 2$ ,  $3 \times 3$ , and  $4 \times 4$  jets and staggered five-jet and 13-jet arrays) were analyzed for the area-averaged heat transfer coefficient ( $h_{avg}$ ),  $\Delta T_{max}$ , and pressure drop at a constant mass flow rate of  $1.33 \times 10^{-4} \text{ kg s}^{-1}$  for  $d/H = 0.25$ ,  $d/H = 0.33$  and  $d/H = 0.5$  (Fig. 6). At a constant mass flow rate, increasing the number of jets led to decreasing the mass flow rate per jet and the jet Reynolds number. In general, increasing the number of jets decreased the  $\Delta T_{max}$  except for the increase in number of jets from 4 to 5 (Fig. 6(a)) where interjet spacing increased significantly even with the increase in the number of jets as shown in Figure 2's configuration. The increase in interjet spacing developed high temperature zones between the adjacent jets. The heat transfer coefficient and the pressure-drop decreased consistently with an increase in the number of jets, showing the weakened effects of stagnation heat transfer due to the decrease in jet velocity for all parameters explored in the present study (Figs. 6(b) and (c)). The increase in the number of jets significantly decreased the  $\Delta T_{max}$  and the pressure drop.

The inline  $4 \times 4$  jet array design was selected for further analysis of performance parameters. The designs with lower  $\Delta T_{max}$  were achieved with higher pressure-drop penalties (Fig. 7). In view of the limited pumping powers available at the micro-level and to maintain flow within the laminar regime, the range of the mass flow rates was set to vary from  $1.33 \times 10^{-4}$  to  $7.98 \times 10^{-4} \text{ kg s}^{-1}$ . The  $\Delta T_{max}$  and pressure-drop change nonlinearly with the  $d/H$  ratio and mass flow rate (Fig. 7(c)).

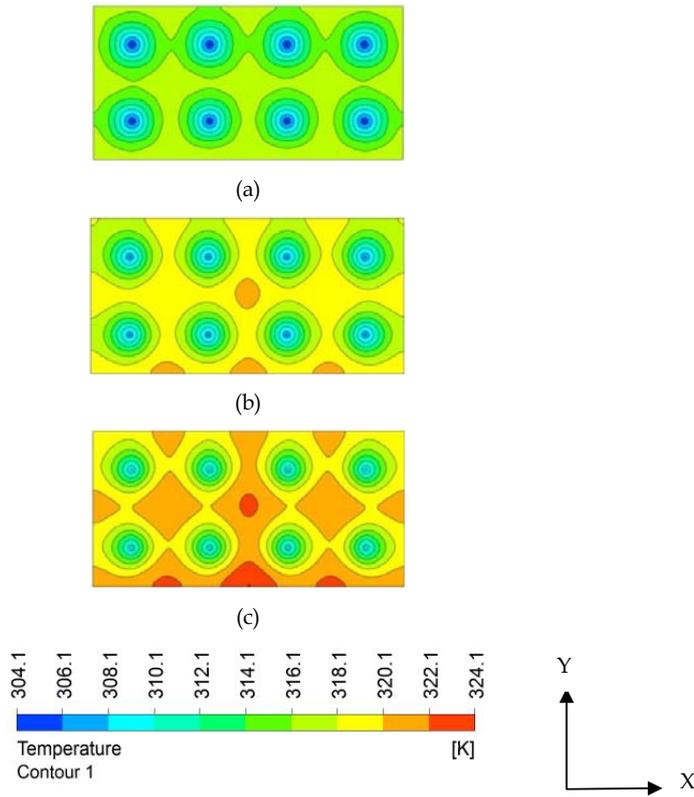
The higher jet velocities, which stem from the increased mass flow rate, increased the local as well as the area-averaged heat transfer coefficients, which finally decreased the  $\Delta T_{max}$ . With increasing



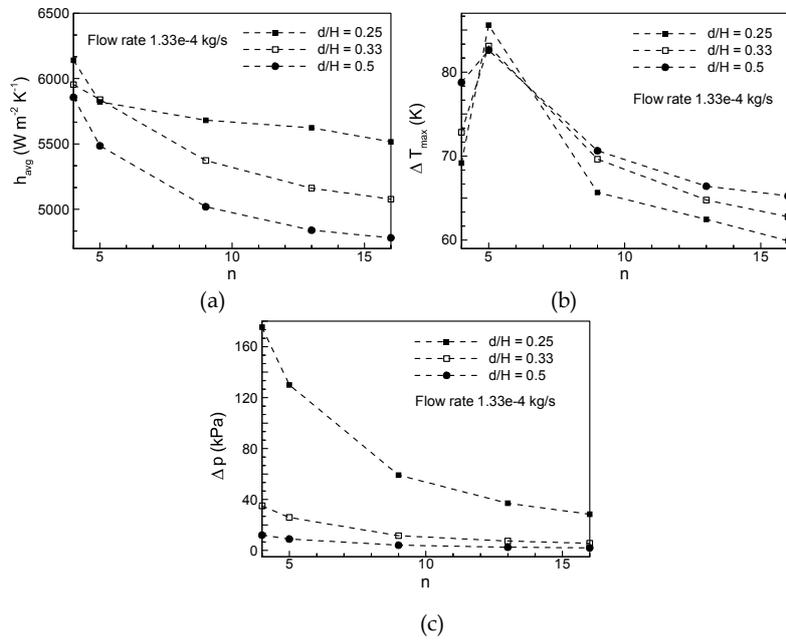
**Figure 4.** Temperature distribution and uniformity ( $-\sigma$ ) on the jet impingement surface for various microjet configurations at  $d/H = 0.5$  and a mass flow rate of  $1.33 \times 10^{-4}$ : (a) Inline  $2 \times 2$  jet array,  $-\sigma = -9.81$ , (b) Staggered five-jet array,  $-\sigma = -13.37$ , (c) Inline  $3 \times 3$  jet array,  $-\sigma = -6.6$ , (d) Staggered 13-jet array,  $-\sigma = -5.24$ , and (e) Inline  $4 \times 4$  jet array,  $-\sigma = -3.4$ .

coolant flow rates, the  $\Delta T_{max}$  decreased and pressure drop increased consistently, but higher coefficient of performance (COP) values were associated with the low flow rates (Table 1). Although an increase of mass flow rate decreased the COP of the model, it improved the temperature uniformity ( $-\sigma$ ) at the heated surface. Table 1 highlights the conflicting nature of the COP and the

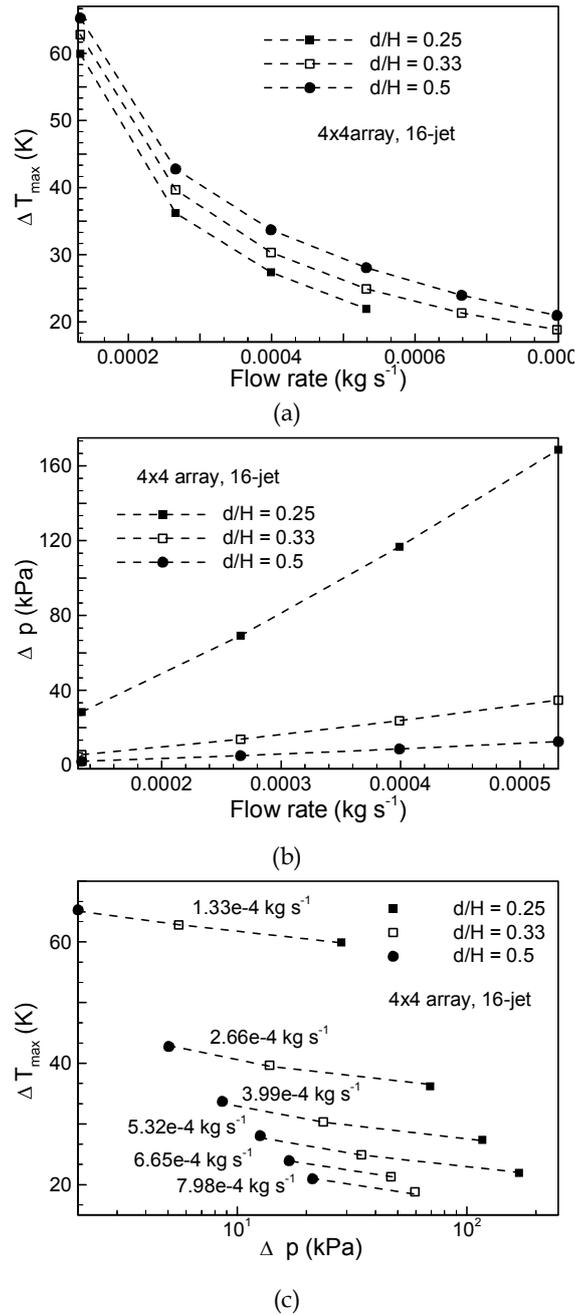
temperature uniformity with the increasing mass flow rate for an inline  $4 \times 4$  jets array. The functional characteristics between the  $\Delta T_{max}$  and pressure drop showed that a higher  $d/H$  ratio gives a lower maximum-temperature rise, lower temperature uniformity and pressure drop but a higher COP, making it advantageous within the range investigated in this study.



**Figure 5.** Temperature distribution and uniformity ( $-\sigma$ ) on the jet impingement surface for inline  $4 \times 4$  jet array at a mass flow rate of  $5.32 \times 10^{-4}$  kg/s. (a)  $d/H = 0.25$ ,  $-\sigma = -2.57$ , (b)  $d/H = 0.33$ ,  $-\sigma = -2.71$ , and (c)  $d/H = 0.5$ ,  $-\sigma = -2.66$ .



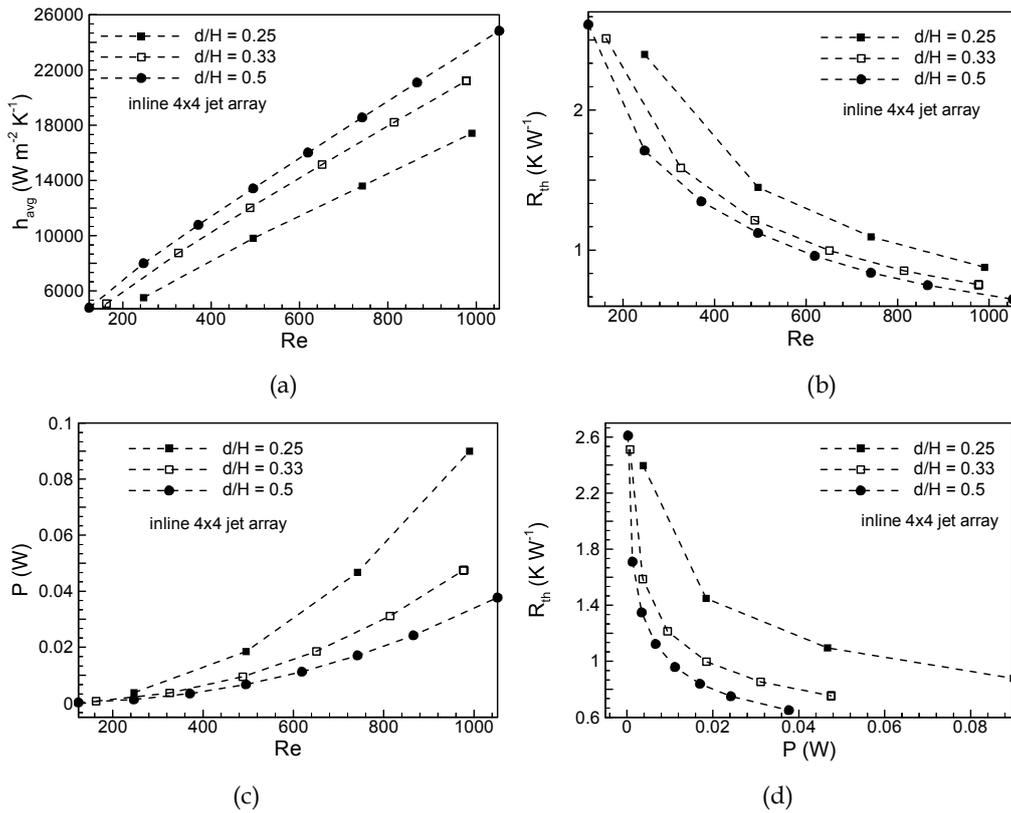
**Figure 6.** Effect of the micro-jet configuration and number of jets on thermal performance at a mass flow rate of  $1.33 \times 10^{-4}$  kg/s and  $d/H = 0.25$ – $0.5$ . (a) Area-averaged heat transfer coefficient ( $h_{avg}$ ), (b) Maximum temperature-rise ( $\Delta T_{max}$ ), and (c) Pressure drop ( $\Delta p$ ).



**Figure 7.** Effect of the mass flow rate on the maximum temperature rise ( $\Delta T_{max}$ ) and pressure drop ( $\Delta p$ ) characteristics for an inline  $4 \times 4$  jet array configuration. (a) Maximum temperature rise ( $\Delta T_{max}$ ), (b) Pressure drop ( $\Delta p$ ), and (c) Maximum temperature rise ( $\Delta T_{max}$ ) versus pressure drop ( $\Delta p$ ).

The jet impingement flow field can be assumed to be laminar for a jet Reynolds number  $< 1000$  (Zuckerman and Lior 2006). The increase in the jet Reynolds number caused an increase in the stagnation heat transfer coefficient, which consequently increased the area-averaged heat

coefficient of the jet impingements. The area-averaged heat transfer coefficient increased monotonously with the increasing jet Reynolds number (Fig. 8(a)) for an inline  $4 \times 4$  jet array with  $d/H = 0.33-0.5$ , resulting in a consistent decrease in the thermal resistance ( $R_{th}$ ). The pressure drop



**Figure 8.** Characteristic variation of area-averaged heat transfer coefficient ( $h_{avg}$ ), thermal resistance ( $R_{th}$ ), and pumping power ( $P$ ) with Reynolds number ( $Re$ ). (a) Area-averaged heat transfer coefficient ( $h_{avg}$ ) versus Reynolds number ( $Re$ ), (b) Thermal resistance ( $R_{th}$ ) versus Reynolds number ( $Re$ ), (c) Pumping power ( $P$ ) versus Reynolds number ( $Re$ ), and (d) Thermal resistance ( $R_{th}$ ) versus pumping power ( $P$ ).

**Table 1.** Maximum temperature rise ( $\Delta T_{max}$ ), coefficient of performance (COP), and temperature uniformity ( $-\sigma$ ) of a multiple micro-jet-impingements cooling model with an inline  $4 \times 4$  jet array at various mass flow rates.

Mass flow rate ( $kg s^{-1}$ )	$\Delta T_{max}$ (K)			COP			$-\sigma$ (K)		
	$d/H = 0.25$	$d/H = 0.33$	$d/H = 0.5$	$d/H = 0.25$	$d/H = 0.33$	$d/H = 0.5$	$d/H = 0.25$	$d/H = 0.33$	$d/H = 0.5$
1.33e-4	59.92	92.79	65.26	6603	33672	92439	-3.76	-3.58	-3.40
2.66e-4	36.21	39.68	42.76	1355	6777	18607	-3.15	-3.09	-2.96
3.99e-4	27.38	30.34	33.70	535	2636	7251	-2.85	-2.87	-2.79
5.32e-4	21.95	24.92	28.07	278	1349	3725	-2.57	-2.71	-2.66
6.65 e-4	-	21.32	23.95	-	803	2229	-	-2.57	-2.54
7.98 e-4	-	18.84	20.96	-	525	1468	-	-2.45	-2.44

$\Delta T_{max}$  = maximum temperature rise; COP = coefficient of performance;  $-\sigma$  = temperature uniformity.

increased consistently with the increase in the Reynolds number due to the higher friction losses in the impingement system, consequently resulting in a higher pumping power ( $P$ ) to drive the fluid through the domain (Fig. 8(c)). The velocity of the fluid passing through the nozzles increased with the decrease in  $d/H$  ratio at the constant jet Reynolds number, which resulted in an increase in the pressure drop and, consequently, an increase in pumping power (Fig. 8(c)). On the other hand, higher jet velocities increased the stagnation as well as area-averaged heat transfer coefficient (Martin 1997), which consequently reduced the convective thermal resistance of the system. The two performance parameters of thermal resistance and pumping power showed a conflicting nature with the change of the design variable or the Reynolds numbers (Fig. 8(d)). The information in Fig. 8(d) suggests that lower thermal resistance can be obtained at the expense of higher pumping power and vice versa. The optimum designs can be obtained from the trade-off curves based on the available pumping power and required thermal resistance.

#### 4. Summary and Conclusion

This study examined the thermal and pressure-drop characteristics of a silicon-based micro-jet impingement cooling model for electronic cooling through 3D numerical analyses. The different micro-jet configurations (*ie.* inline  $2 \times 2$ ,  $3 \times 3$ , and  $4 \times 4$  jets and staggered five-jet and 13-jet arrays with various jet diameter-to-channel height ratios) were examined to determine the temperature and pressure drop, heat transfer coefficient, thermal resistance and pumping power characteristics for a range of flow rates and Reynolds numbers. The convection and radiation at the outer walls were neglected to maintain a conservative thermal analysis. The multiple impingements produced higher temperature uniformity on the heated surface for the jets of all diameters. For the constant flow rates across the domain, the increase in jet diameter increased the maximum-temperature rise in the substrate. In general, the temperature rise and pressure drop decreased with an increase in number of jets. The increase in flow rates, though, increased the temperature uniformity but decreased the coefficient of performance of the model. The thermal resistance and pumping power showed a conflicting nature with the design parameters and jet Reynolds numbers.

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